

### Avionics Questionnaire

Operator: \_\_\_\_\_ Aircraft Model: \_\_\_\_\_ Airplane SN: \_\_\_\_\_

**A. Please complete the following table with the information for the avionics/system components currently installed in the airplane. If other avionics components relevant to the altimetry systems are installed on the aircraft, please include their information in the blank spaces.**

Component	Manufacturer Model	Part Number
#1 Air Data Computer (or Unit)		
#2 Air Data Computer (or Unit)		
#1 Configuration Module [CM1]		
#2 Configuration Module [CM2]		
#1 Altimeter [ALT1]		
#2 Altimeter [ALT2]		
#1 Mach/Airspeed Indicator		
#2 Mach/Airspeed Indicator		
LHS Pitot-Static Probe		
RHS Pitot-Static Probe		
#1 Transponder [XPDR1]		
#2 Transponder [XPDR2]		
Altitude Alerter		
Autopilot System		
Autopilot Amplifier		
Autopilot Computer		
Autopilot Panel/Controller		
#1 Air Data Control (if applicable)		
#2 Air Data Control (if applicable)		
Air Data Sensor (if applicable)		

Table 1. Avionics Component Information for RVSM Operations

**Please provide the following information:**

- a) Operating Empty Weight (OEW): \_\_\_\_\_ lbs.
- b) On a short flight, cruising @ FL290, what is the lowest gross weight the aircraft would be expected to operate? \_\_\_\_\_ lbs.
- c) What is the maximum Mach or airspeed at which typical operations might be conducted?

**With your completed questionnaire, please attach an air data/altimetry system block diagram and a Pitot-static system schematic (normally found in the Maintenance Manual).**

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**B. Please answer the following questions about the airplane configuration:**

- 1) Suppose the #1 ADC (ADC1<sup>1</sup>) fails during normal flight.
  - a) What annunciation will be provided to the crew (i.e. how will they know)?
  - b) If the autopilot is flying off the #1 system, will the autopilot disengage? **Yes\_\_ No\_\_**
  - c) Will the altitude alerter function normally? **Yes\_\_ No\_\_**
    - i) If not, what action needs to be taken to ensure the altitude alerter functions normally?
  
- 2) Suppose the #2 ADC (ADC2<sup>1</sup>) fails during normal flight.
  - a) What annunciation will be provided to the crew (i.e. how will they know)?
  - b) If the autopilot is flying off the #2 system, will the autopilot disengage? **Yes\_\_ No\_\_**
  - c) Will the altitude alerter function normally? **Yes\_\_ No\_\_**
    - i) If not, what action needs to be taken to ensure the altitude alerter functions normally?
  
- 3) Is there an "ADC Select" switch installed in the airplane? **Yes\_\_ No\_\_**
  
- 4) What type of Pitot static system is installed on the aircraft (Fill in the blanks)
  - a) Pitot probes and flush mounted static sources \_\_\_\_\_
  - b) Pitot static probes only. \_\_\_\_\_ How many? \_\_\_\_\_
  - c) A combination of both. Please explain.
  
- 5) Does the airplane have a static source select switch, allowing the crew to select the primary (normal) or standby (alternate) static sources? **Yes\_\_ No\_\_**
  - a) If so, is there a single switch, or one each for the Pilot and CoPilot?  
**Single Switch\_\_ There is one for the Pilot & one for the CoPilot\_\_**
  
- 6) Is there a transponder select switch installed on the airplane, allowing the flight crew to select which transponder is reporting to ATC? **Yes\_\_ No\_\_**
  - a) If so, is it manual or automatic? **Manual\_\_ Automatic\_\_**
  - b) Does the transponder select switch also select the air data system (#1 or #2) from which altitude information is obtained? **Yes\_\_ No\_\_**
  
- 7) To which system (#1/Pilot or #2/CoPilot) is the altitude alerter connected? **#1\_\_ #2\_\_ Both\_\_ Neither\_\_**
  - a) Can either system be selected to provide data to the altitude alerter? **Yes\_\_ No\_\_ N/A\_\_**
  - b) While cruising with the autopilot in altitude hold mode, at what height from the cruise flight level will the altitude alerter generate an aural warning (i.e. what is the altitude alert threshold of the altitude alerter in altitude deviation mode?) \_\_\_\_\_ **feet.**
  
- 8) Is the airplane equipped with a separate air data sensor, computer or other pressure-sensing unit that provides altitude and/or static pressure information to the autopilot while flying in altitude hold mode?  
**Yes\_\_ No\_\_**
  - a) What happens to the autopilot if this separate computer/sensor unit should fail (i.e. autopilot dis-engages, a second system takes over automatically or manually, etc.)?
  
- 9) Have there been any Service Bulletins or other modifications implemented, on the aircraft, which affect any of the aircraft systems identified above or the performance of the aircraft at cruise? If so, please explain on a separate sheet and attach to this survey.

<sup>1</sup>ADC refers to primary air data system component installed on the aircraft (i.e. Air Data Computer, Air Data Sensor, Air Data Unit, Air Data Display Unit, etc.).